DESIGN DEVELOPMENT AND RAPID PROTOTYING OF RUBBER COMPONENTS USING FEA

By

KARTIK SRINIVAS



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Education

B.E. Mechanical Eng.
 L.D. College of Engineering -1998

M.S. Mechanical Eng.
 Wright State University - 2001

Polymer Eng. Course Certifications Univ. of Akron – 2002-2003

– CNC Machinist Certificate
 – Akron Machining Institute - 2004

Professional Experience

- Mechanical Testing of Engineering Materials: MTS®, Instron® and Proprietary Servo-hydraulic and Electro-mechanical Load frames.
- Performance Characterization of Elastomeric Products.
- Finite Element Analysis of Engineering Products and Components
 - Automotive (Under the hood, suspension components, tires, etc.).
 - Biomedical (Spinal, knee, and hip implants, stents etc.)
 - Aerospace (Static and dynamic analysis for honeycomb structures etc.)
- Durability Testing and Lifetime Prediction using Accelerated Test Conditions.

NABL ISO/IEC 17025:2017 ACCREDITED LAB





National Accreditation Board for Testing and Calibration Laboratories

NABL

CERTIFICATE OF ACCREDITATION

ADVANCED SCIENTIFIC AND ENGINEERING SERVICES (ADVANSES)

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"General Requirements for the Competence of Testing & Calibration Laboratories"

for its facilities at

PLOT NO 49, MOTHER INDUSTRIAL PARK, ZAK-KADADARA ROAD, NEAR ZAK GIDC, AHMEDABAD, GUJARAT, INDIA

in the field of

TESTING

Certificate Number:

TC-9168

Issue Date:

22/12/2020

Valid Until:

21/12/2022

This certificate remains valid for the Scope of Accreditation as specified in the annexure subject to continued satisfactory compliance to the above standard & the relevant requirements of NABL.

(To see the scope of accreditation of this laboratory, you may also visit NABL website www.nabl-india.org)

Signed for and on behalf of NABL



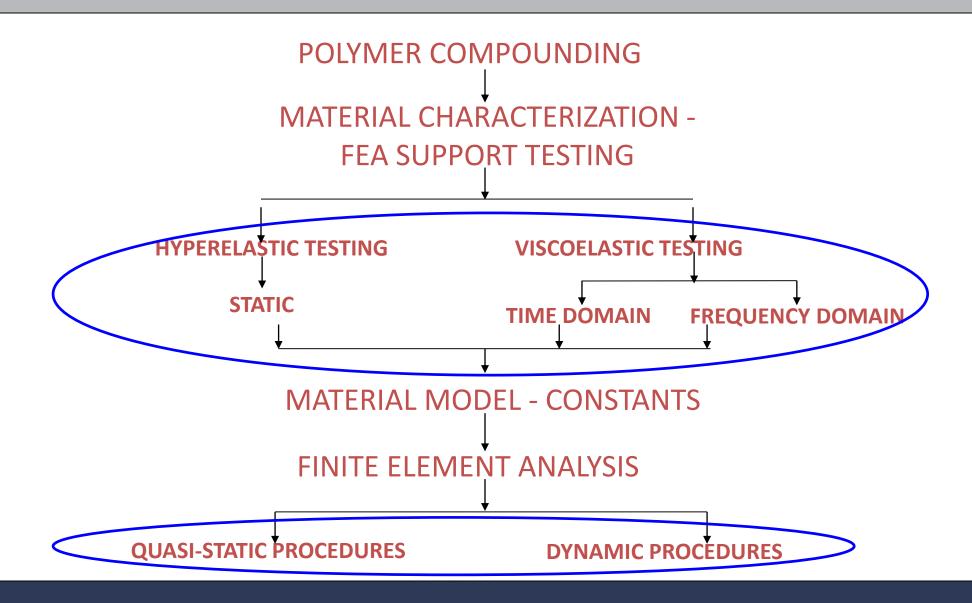


N. Venkateswaran Chief Executive Officer

Accredited for;

- 1. Static Testing
- 2. Dynamic Testing
- 3. Fatigue Testing

POLYMER ANALYSIS METHODOLOGY



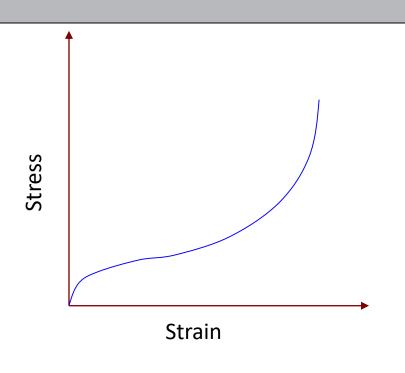
COMPLEXITIES IN THE FEA OF POLYMERS

- The Stress-Strain relationship is highly non-linear.
- Strain range can vary from 5 % to 700 %.
- Imperative to use large deflection theory.
- Stress-Strain characteristics are temperature dependent and hysteresis effects are significant.
- Rate effects are highly pronounced.
- Too many material models using different approaches.

MAJOR FEA CODES IN THE MARKET

- Abaqus® Implicit and Explicit Rubber
- •Ansys® Implicit Rubber
- Calculix Implicit and Explicit Rubber
- •Comsol® Implicit—Rubber (50%)
- •LS-Dyna® Implicit and Explicit Rubber
- •MSC Marc-Nastran® Implicit and Explicit Rubber

HYPERELASTIC CONSTITUTIVE MODELS



Stretch Ratio (λ)

 $\lambda = 1 + Strain = 1 + \varepsilon$

Input:

Stress-Strain Data from Main Deformation modes

W = Constants(
$$I_1(\lambda)$$
, $I_2(\lambda)$, $I_3(\lambda)$)

W = C10, C01,
$$\alpha_{i_1} \mu_i (I_1(\lambda), I_2(\lambda), I_3(\lambda))$$

$$I_{1} = \lambda_{1}^{2} + \lambda_{2}^{2} + \lambda_{3}^{2}$$

$$I_{2} = \lambda_{1}^{2} \lambda_{2}^{2} + \lambda_{2}^{2} \lambda_{3}^{2} + \lambda_{3}^{2} \lambda_{1}^{2}$$

$$I_{3} = \lambda_{1}^{2} \lambda_{2}^{2} \lambda_{3}^{2}$$

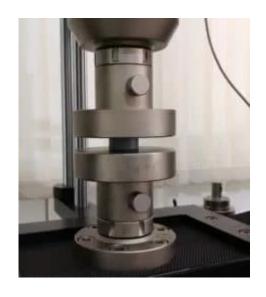
HYPERELASTIC MATERIAL CHARACTERIZATION TESTING

- 1. Five Deformation Modes
 - Uniaxial tension
 - Uniaxial compression
 - Planar shear
 - Volumetric compression
 - Equibiaxial tension
- 2. Testing speed = quasi-static procedure
- 3. Minimum of 3 specimens



Uniaxial Tension Test

$$\lambda_1 = \lambda_T, \lambda_2 = \lambda_3 = \lambda_T^{-1/2}$$

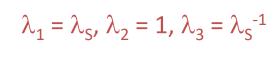


Uniaxial Compression Test

$$\lambda_1 = \lambda_C, \lambda_2 = \lambda_3 = \lambda_C^{-1/2}$$



Planar Shear Test





Volumetric Compression Test

$$\lambda_1 = \lambda_2 = \lambda_3 = \lambda_V$$



Equibiaxial Tension Test

$$\lambda_1 = \lambda_2 = \lambda_B$$
, $\lambda_3 = 1/\lambda_B^2$

STRESS AND STRAIN MEASURES

2nd Piola Kirchoff stress – Cauchy's stress

Cauchy's stress = Force / Final area

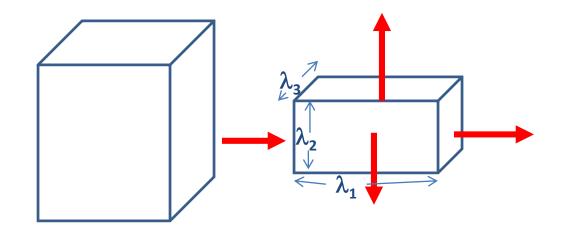
2nd Piola Kirchoff stress = Force / Original area

Green Strain - True strain

True Strain = dL/L

Green Strain = dL/L + Nonlinear terms

The deformation is described by stretch ratios: $\lambda_1, \lambda_2, \lambda_3$

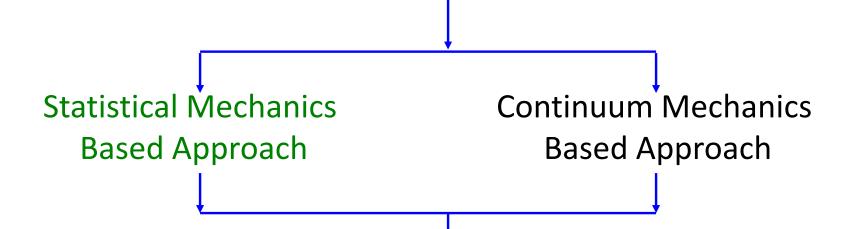


Strain Energy W

A simple relation for the strain energy density is:

W= (E/6)
$$(\lambda_1^2 + \lambda_2^2 + \lambda_3^2 - 3)$$
 ----- 1

MATERIAL MODELS



- 1. Neo-Hooke
- 2. Mooney-Rivlin
- 3. Polynomial
- 4. Ogden
- 5. Van der Waals
- 6. Yeoh
- 7. Arruda-Boyce

- 8. Gent
- 9. Peng
- 10. Peng-Landel
- 11. Klesner-Segal
- 12. Hart-Smith
- 13. Valanis-Landel
- 14. User Defined

STRESS - STRAIN RELATIONS

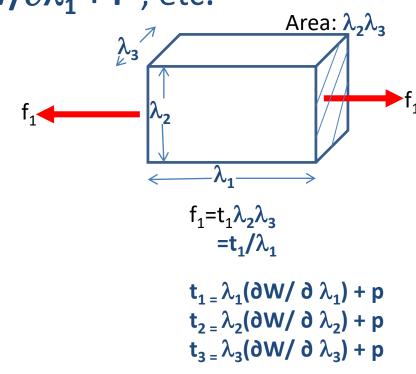
We can calculate stresses from the relation for W.

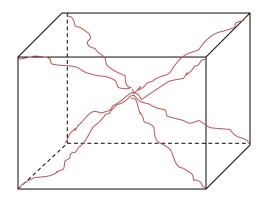
$$\Delta W = f_1 d\lambda_1$$
. Hence: $\mathbf{t_1} = \lambda_1 \partial \mathbf{W}/\partial \lambda_1 + \mathbf{P}$, etc.

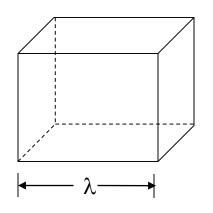
where P is an undetermined hydrostatic pressure

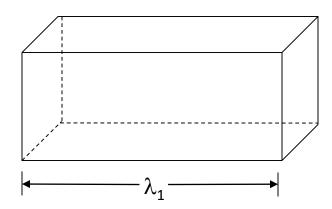
Note: we have assumed incompressibility:

$$\lambda_1 \lambda_2 \lambda_3 = 1$$









8- Chain Network

Deformation in a Continuum Element

$$W(I_1, I_2, I_3) = \sum_{IJK=0}^{N} C_{ijk} (I_1 - 3)^i (I_2 - 3)^j (I_3 - 3)^k + \sum_{i=1}^{N} \frac{1}{D_i} (J^{el} - 1)^{2i}$$

Neo-Hookean Model: $W = C_{10}(I_1 - 3)$

Mooney-Rivlin Model:
$$W = C_{10}(I_1 - 3) + C_{01}(I_2 - 3)$$

Ogden Model:
$$W = \sum_{i=1}^{N} \frac{2\mu_i}{\alpha_i^2} (\lambda_1^{-\alpha_i} + \lambda_2^{-\alpha_i} + \lambda_3^{-\alpha_i} - 3)$$

VISCOELASTIC MATERIAL CHARACTERIZATION TESTING

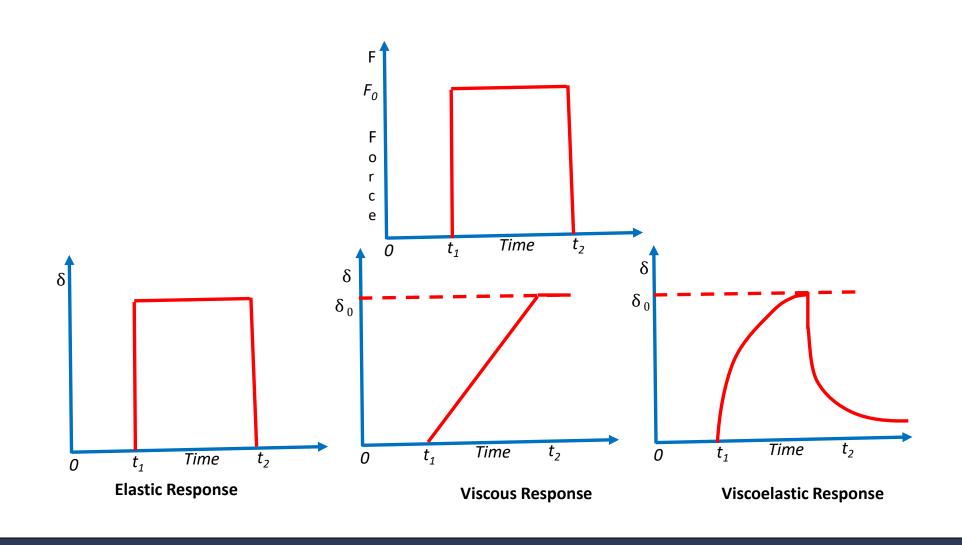
TIME DOMAIN VISCOELASTIC TESING

- STRESS RELAXATION
- CREEP

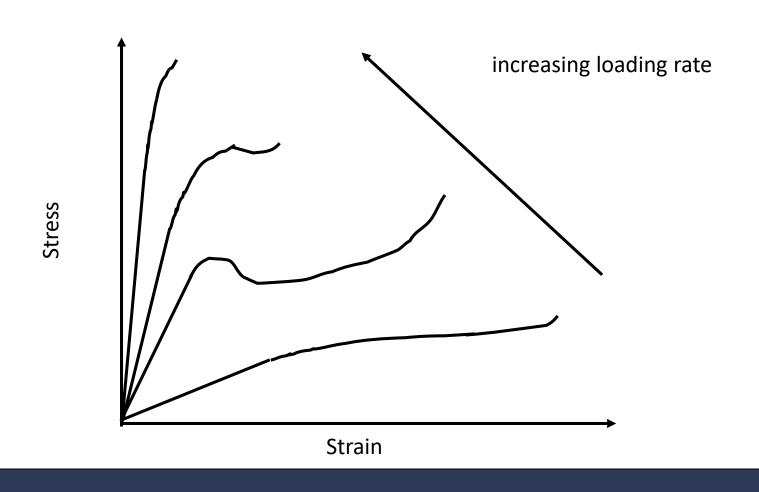
FREQUENCY DOMAIN VISCOELASTIC TESING

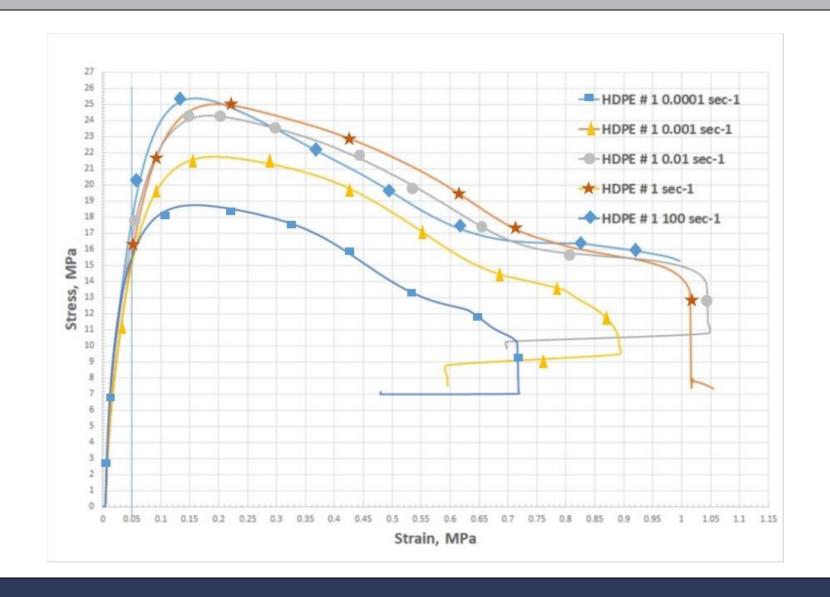
- STRAIN SWEEPS
- FREQUENCY SWEEPS

APPLIED LOAD AND RESULTANT DEFORMATION



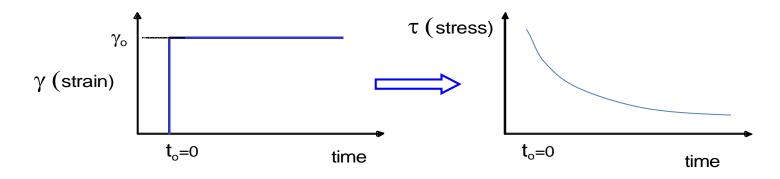
LOADING RATE VS. MODULUS





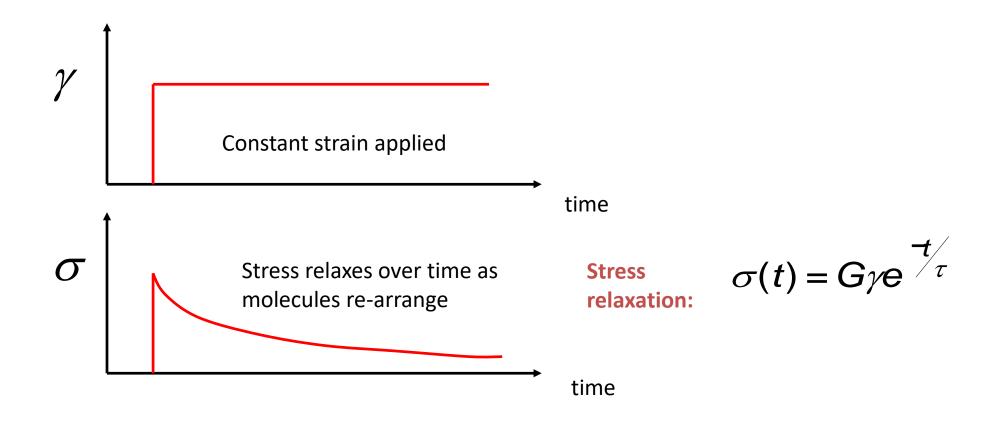
VISCOELASTICITY AND STRESS RELAXATION

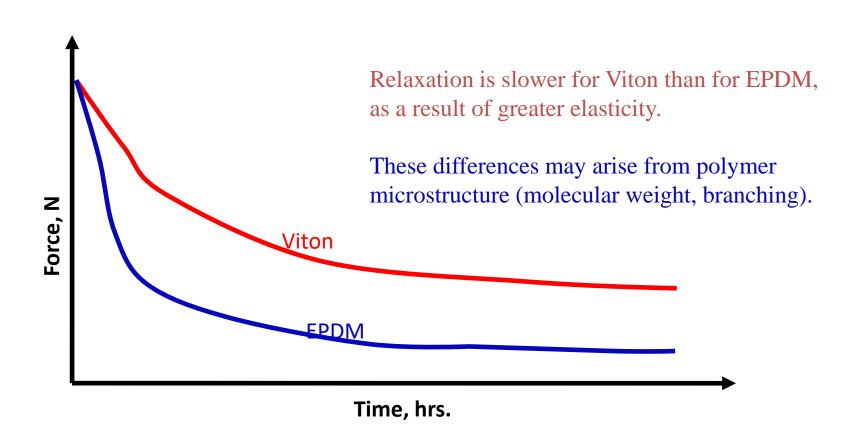
Whereas steady-shear measurements probe material responses under a steady-state condition, creep and stress relaxation monitor material responses as a function of time.



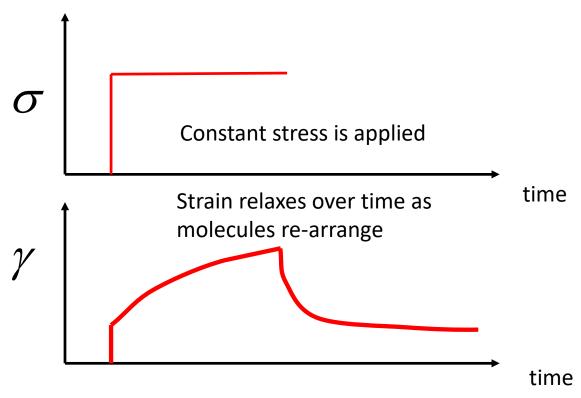
Stress relaxation studies the effect of a step-change in strain on stress.

PHYSICAL MEANING OF THE RELAXATION TIME





CREEP



Constant stress is applied and the strain relaxes as function of time

MATERIAL TESTING

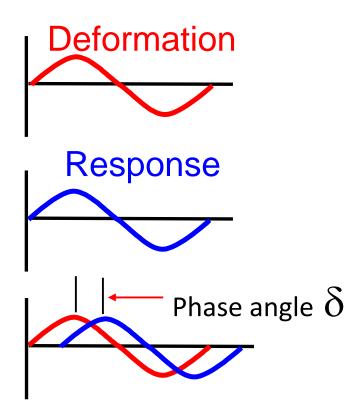
- MODULUS (STATIC/DYNAMIC)
- TAN δ
- CRACK GROWTH **PROPAGATION**



Courtesy: Dynamic Testing Equipments

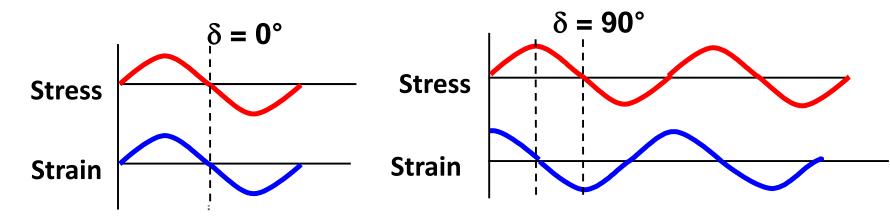
DYNAMIC MECHANICAL TESTING

- An oscillatory (sinusoidal) deformation (stress or strain) is applied to a sample.
- The material response (strain or stress) is measured.
- •The phase angle δ , or phase shift, between the deformation and response is measured.

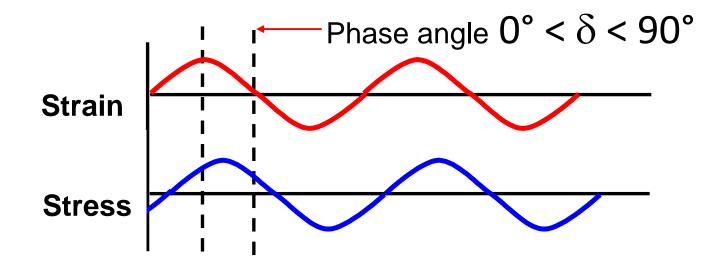


Purely Elastic Response (Hookean Solid)

Purely Viscous Response (Newtonian Liquid)



DYNAMIC MECHANICAL TESTING: VISCOELASTIC MATERIAL RESPONSE



Viscoelastic Parameters: The Complex, Elastic, & Viscous Stress

- The stress in a dynamic experiment is referred to as the complex stress σ^*
- ➤ The complex stress can be separated into two components:
 - 1) An elastic stress in phase with the strain. $\sigma' = \sigma^* \cos \delta$ σ' is the degree to which material behaves like an elastic solid.
 - 2) A viscous stress \Box in phase with the strain rate. $\sigma'' = \sigma^* \sin \delta$ σ'' is the degree to which material behaves like an ideal liquid.



VISCOELASTIC PARAMETERS

The Modulus: Measure of materials overall resistance to deformation.

G = Stress/Strain

The Elastic (Storage) Modulus:
Measure of elasticity of material. The ability of the material to store energy.

 $G' = (stress/strain)cos\delta$

The Viscous (loss) Modulus:

The ability of the material to dissipate energy. Energy lost as heat.

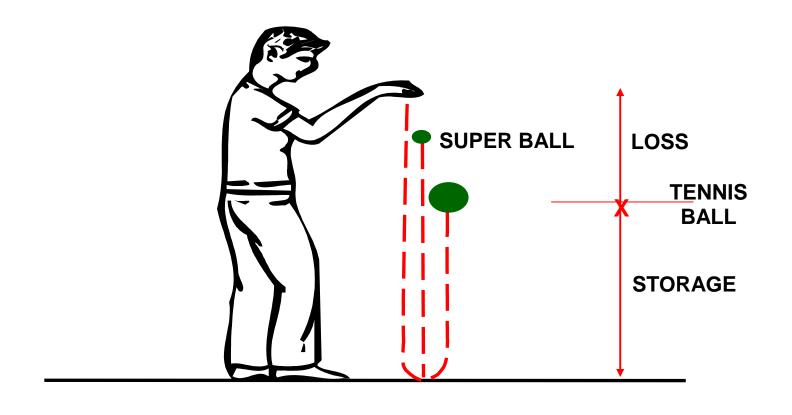
 $G'' = (stress/strain)sin\delta$

Tan Delta:

Measure of material damping - such as vibration or sound damping.

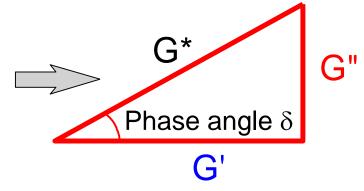
Tan δ = G"/G'

STORAGE AND LOSS OF A VISCOELASTIC MATERIAL



VISCOELASTIC PARAMETERS: DAMPING, TAN δ

Dynamic measurement represented as a vector

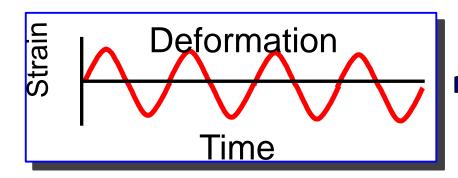


 The tangent of the phase angle is the ratio of the loss modulus to the storage modulus.

tan
$$\delta = G''/G'$$

ullet "TAN DELTA" (tan δ) is a measure of the damping ability of the material.

DYNAMIC TIME SWEEP



The material response
 is monitored at a constant frequency, amplitude and temperature.

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Multi-Frequency

METHOD

Equilibrate at ____°C

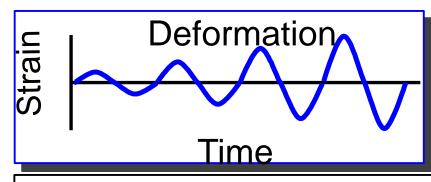
Isotherm for ___ min.

Frequency = single/multiple

Amplitude = In Linear viscoelastic region
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- USES
- ∠Cure Studies

DYNAMIC STRAIN SWEEP



AINSTRUMENT MODE

Multi-Strain

Equilibrate at ____°C

Isotherm for ____ min.

Strain Sweep

▲Frequency = single only

▲Amplitude = Program Table up to 28 values

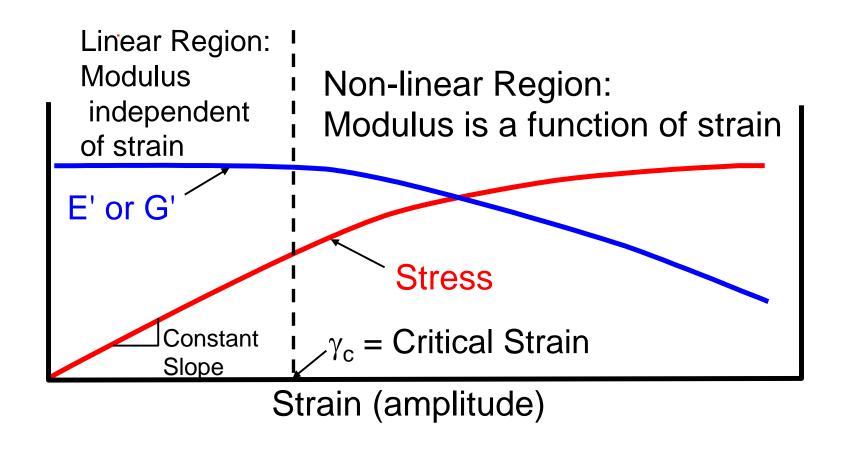
 The material response to increasing deformation amplitude is monitored at a constant frequency and temperature.

•USES

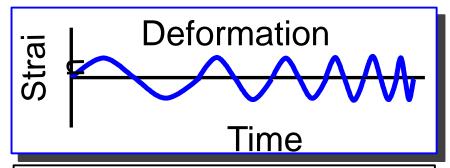
∠Identify Linear Viscoelastic Region

∠Resilience

DYNAMIC STRAIN SWEEP: MATERIAL RESPONSE



FREQUENCY SWEEP



AINSTRUMENT MODE

Multi-Frequency

METHOD

Equilibrate at ____°C

Isotherm for min.

Frequency Sweep

▲Frequency = Program Table up to 28

values

<u>∧Amplitude</u> = single only In Linear

viscoelastic region

 The material response to increasing frequency (rate of deformation) is monitored at a constant amplitude and temperature.

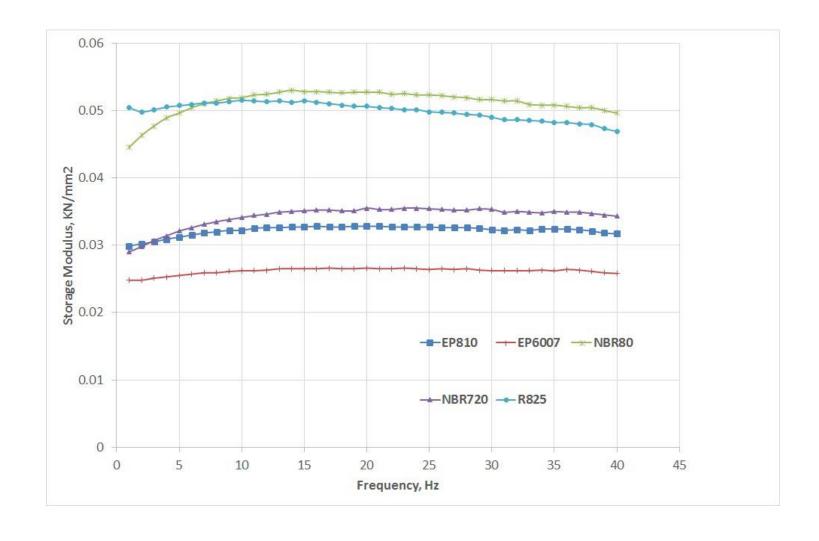
•USES

- ∠High and Low Rate (short and long time) modulus properties.
- ∠Polymer melt processing (shear sandwich).
- ∠Extend range with TTS

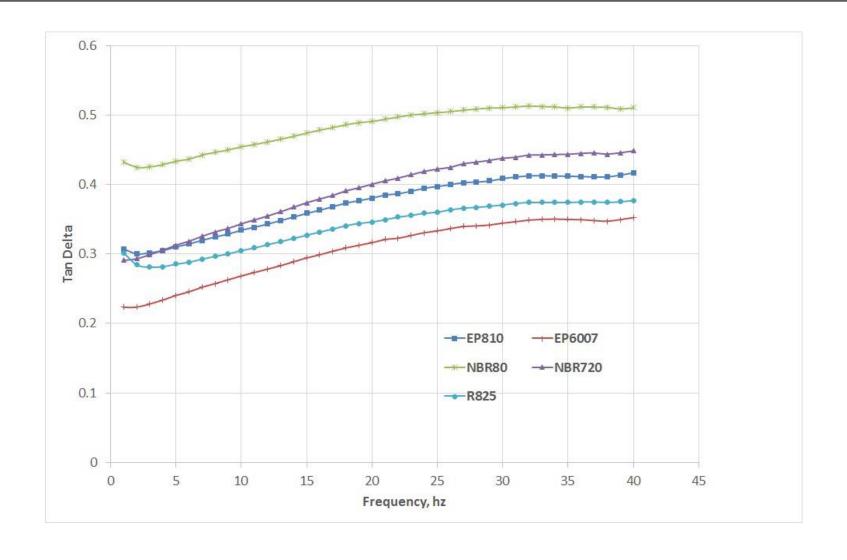
MATERIAL CHARACTERIZATION DMA TEST DATA

KS									
Step	Fundamental	Dynamic	Damping	Delta	Phase	Complex	Storage	Loss	TanDelta
No.	Frequency	Stiffness	Damping	Deita	Huse		Modulus		
	yucy						1110 01011010		
						kN/mm^	kN/mm^	kN/mm^	
	Hz	kN/mm	kNs/mm	Degrees	rad	2	2	2	
1	1	1.66425	0.05775	14.09262	0.245963	0.04953	0.04836	0.010725	0.29016
2	2	1.66575	0.028875	14.1101	0.246268	0.04953	0.04836	0.010725	0.2912
3	3	1.685125	0.019875	14.38483	0.251063	0.050115	0.048945	0.011115	0.299
4	4	1.702	0.015375	14.73293	0.257138	0.0507	0.049335	0.011505	0.300875
5	5	1.717125	0.01275	15.13568	0.264168	0.05109	0.049725	0.011895	0.31239
6	6	1.7335	0.011	15.4504	0.26966	0.051675	0.050115	0.012285	0.31915
7	7	1.746125	0.00975	15.8461	0.276567	0.052065	0.050505	0.012675	0.32773
8	8	1.75425	0.00875	16.16373	0.28211	0.05226	0.050505	0.013065	0.33462
9	9	1.76575	0.008	16.49682	0.287924	0.05265	0.050895	0.013455	0.34177
10	10	1.773	0.007375	16.82195	0.293599	0.052845	0.05109	0.01365	0.34879
11	11	1.7815	0.00675	17.12603	0.298906	0.05304	0.05109	0.01404	0.35542
12	12	1.789375	0.006375	17.40917	0.303847	0.053235	0.051285	0.014235	0.36153
13	13	1.801625	0.006	17.70261	0.308969	0.053625	0.051675	0.014625	0.36803
14	14	1.805875	0.00575	18.03458	0.314763	0.05382	0.051675	0.01482	0.37531
15	15	1.812	0.005375	18.36453	0.320522	0.054015	0.051675	0.01521	
16	16	1.808875	0.005125	18.63523	0.325246	0.05382	0.051675	0.015405	0.38844
17	17	1.82025	0.005	18.93774	0.330526	0.05421	0.05187	0.015795	0.3952
18	18	1.812875		19.21472	0.33536	0.054015	0.051675	0.01599	0.40131
19	19	1.820125	0.0045	19.43917	0.339278	0.05421	0.051675	0.016185	0.40625
20	20	1.824875	0.004375	19.66048	0.34314	0.054405	0.05187	0.01638	0.41119

PLOT OF STORAGE MODULUS VS FREQUENCY FROM A FREQUENCY SWEEP TEST



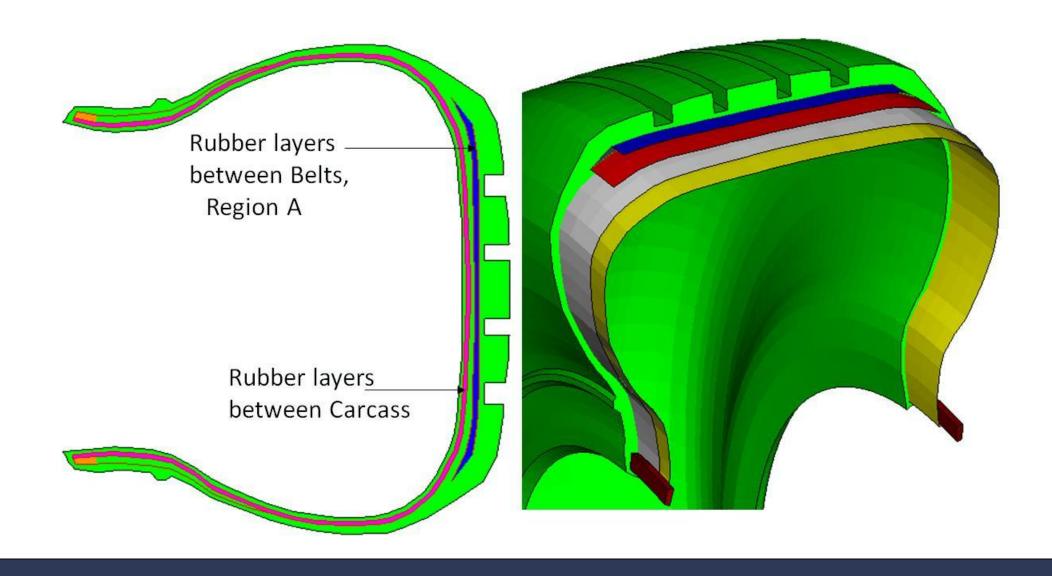
PLOT OF TAN DELTA VS FREQUENCY FROM A FREQUENCY SWEEP TEST



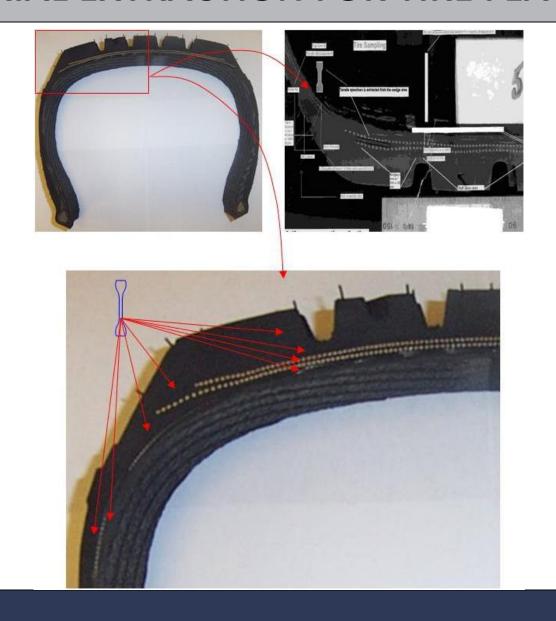
TIME DOMAIN AND FREQUENCY DOMAIN CALCULATIONS ON A DMA TEST DATA TO GENERATE VISCOELASTIC PROPERTIES

gi	ti	Frequency	Comp. Mod.	St. Mod.	Loss Mod									Frequency	Prony ser	ies sum
						Factor A	Factor B	Factor C	G-storage	G-loss	GI / Ginf	1-Gs/Ginf			time	relative modulu
0	0.00001	1	0.0123	0.0121	0.0022	0.5	2.5000000E-01	7.1801780E-02	7.5000000E-01	7.1801780E-02	1.4360356E-01	-5.00000E-01	0	0 1	0	
	0.0001	3.0417	0.0125	0.0123	0.0024		2.9869758E-01	6.4162397E-02	7.9869758E-01	6.4162397E-02	1.2832479E-01	-5.97395E-01	0	0 3.0417	0.00001	0.9998
	0.001	5.0833	0.0128	0.0125	0.0026		3.1702684E-01	6.6567664E-02	8.1702684E-01	6.6567664E-02	1.3313533E-01	-6.34054E-01	0	0 5.0833	0.0001	0.9988
0.1	0.01	7.125	0.013	0.0127	0.0029		3.3222530E-01	6.9655339E-02	8.3222530E-01	6.9655339E-02	1.3931068E-01	-6.64451E-01	0	0 7.125	0.001	0.9893
0.1	0.1	9.1667	0.0136	0.013	0.0031		3.4530574E-01	7.0882270E-02	8.4530574E-01	7.0882270E-02	1.4176454E-01	-6.90611E-01	0	0 9.1667	0.01	0.9261
0.1	1	11.2083	0.01355	0.0131	0.0033		3.5612177E-01	7.0578545E-02	8.5612177E-01	7.0578545E-02	1.4115709E-01	-7.12244E-01	0	0 11.2083	0.1	0.8261
0.1	10	13.25	0.0135	0.0131	0.0035		3.6486380E-01	6.9439429E-02	8.6486380E-01	6.9439429E-02	1.3887886E-01	-7.29728E-01	0	0 13.25	0.2	0.7932
0.1	100	15.2917	0.0136	0.0131	0.0037		3.7189996E-01	6.7979272E-02	8.7189996E-01	6.7979272E-02	1.3595854E-01	-7.43800E-01	0	0 15.2917	0.3	0.7758
0	1000	17.3333	0.0137	0.0132	0.0039		3.7760938E-01	6.6497703E-02	8.7760938E-01	6.6497703E-02	1.3299541E-01	-7.55219E-01	0	0 17.3333	0.4	0.7645
0	10000	19.375	0.014	0.0134	0.0041		3.8231438E-01	6.5145259E-02	8.8231438E-01	6.5145259E-02	1.3029052E-01	-7.64629E-01	0	0 19.375	0.5	0.75599
		21.4167	0.0141	0.0134	0.0042		3.8626635E-01	6.3984961E-02	8.8626635E-01	6.3984961E-02	1.2796992E-01	-7.72533E-01	0	0 21.4167	0.6	0.7487
0.5		23,4583	0.0141	0.01345	0.0042		3.8965495E-01	6.3032297E-02	8.8965495E-01	6.3032297E-02	1.2606459E-01	-7.79310E-01	0	0 23.4583	0.7	0.7422
		25.5	0.0142	0.0135	0.0044		3.9262150E-01	6.2278684E-02	8.9262150E-01	6.2278684E-02	1.2455737E-01	-7.85243E-01	0	0 25.5	0.8	0.7364
G_0= G_inf=	1 0.5		0.90 0.80 0.80 0.80 0.80 0.60 10.50 10.50 10.50 0.40 0.30 0.20	ormalized	d Modul	us Vs. Freq	Storage Modulus	1.2 1 0.8 0.6 0.6 0.4 0.2	Stress Relaxat	ion Vs. Time			Factor A	<i></i>	$\underbrace{+^{G}\tau_{i}^{2}\omega^{2}}_{\text{tor B}}$	
			0.10 0.00 0 5	10	15 20	25 3	0	-0.1 0	0.1 0.2 0.3 0	4 05 0.6 0	7 0.8 0.9		_	$\frac{1 + t_i \omega^2}{\text{Factor C}}$		

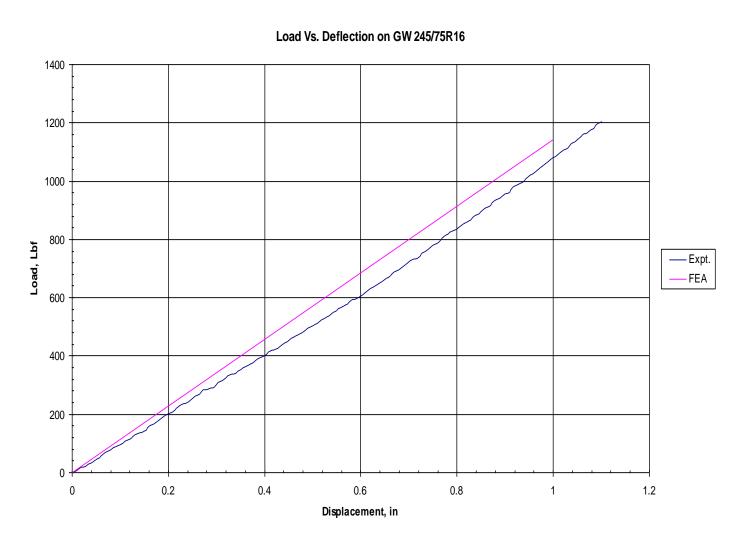
APPLICATION OF HYPERELASTIC & VISCOELASTIC DATA FOR TIRE PERFORMANCE ANALYSIS



MATERIAL EXTRACTION FOR TIRE FEA

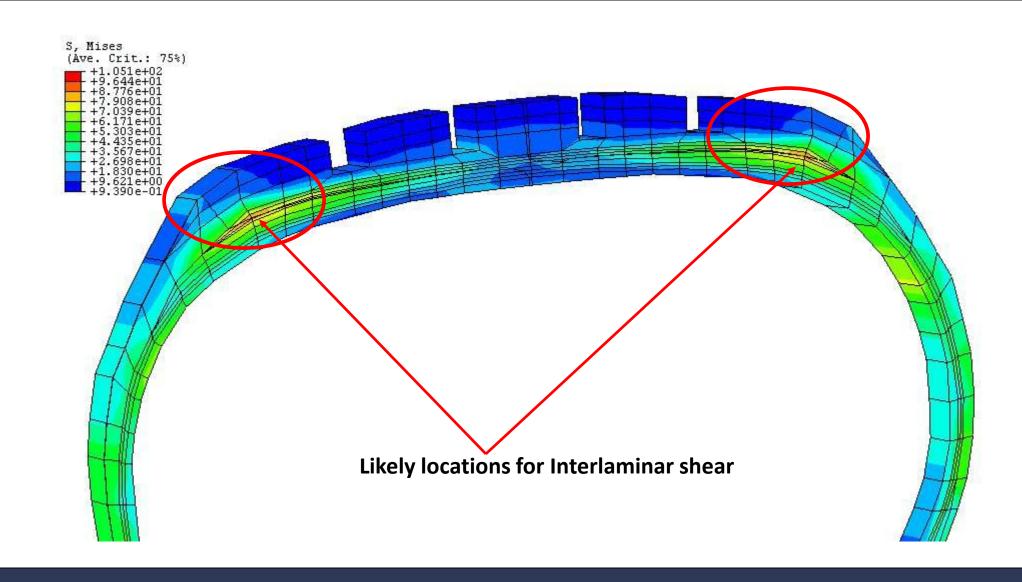


LOAD-DEFLECTION RESULTS



Comparison of Experimental and FEA Results

HYPERELASTIC TIRE ANALYSIS RESULTS

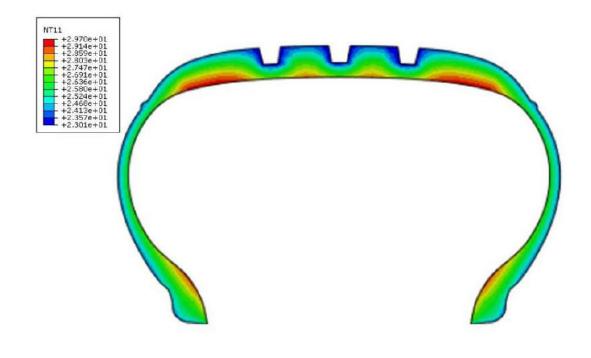


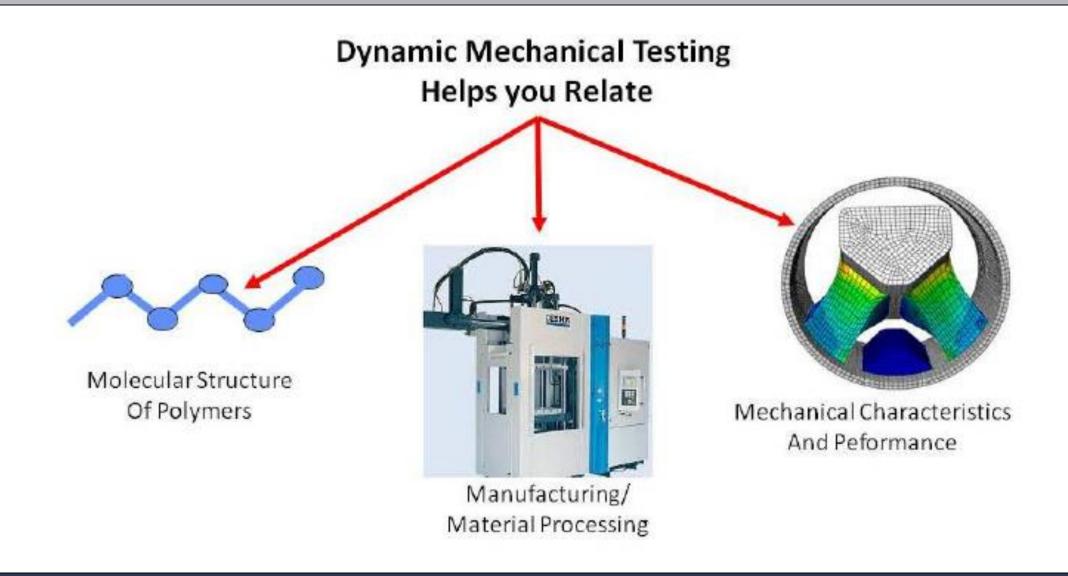
VISCOELASTIC TIRE ANALYSIS RESULTS FOR TEMPERATURE PREDICTION

Non-linear Viscoelastic tire simulation is carried out using Abaqus/Ansys/Marc etc., to predict the hysteresis losses, temperature distribution and rolling resistance of a tire. The simulation includes several steps like (a) FE tire model generation, (b) Material parameter identification, (c) Material modeling and (d) Tire Rolling Simulation.

The energy dissipation and rolling resistance are evaluated by using dynamic mechanical properties like storage and loss modulus, tan delta etc. The heat dissipation energy is calculated by taking the product of elastic strain energy and the loss tangent of materials.

Tire Temperature Prediction during Rolling





VERIFICATIONS AND VALIDATIONS

• Single Element Tests

- Can be performed by analyst (uses curve-fitting)
- First order check on the material property

Closed Loop Validation

- Compare results with the original material data from experiments
- Open Loop Validation
 - Comparison to alternate experimental modes
 - Carry out tests at component level

APPLICATION AREAS IN FEA

- Engine mounts, bushings, vibration isolators etc.
- Seals, o-rings etc.,
- Gaskets(Rubber and Composite)
- Weather strips
- Biomedical implants
- All situations where static pre-load is superimposed by service loads.

GOAL OF CAE TESTING SERVICES

- Create a step-by-step process based procedure for CAE.
- Validate for different regimes of product performance and service conditions
 - Viscoelastic
 - Hyperelastic
 - Hyperelastic rate dependent
 - Non-linear (Elasto-plastic, high strain rate)
- Validate models for failure analysis and design optimization

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HOW CAN FEA HELP THE DESIGN ENGINEER & POLYMER TECHNOLOGIST

- Provides virtual testing and approaches to simulating complex material characteristics.
- use of FEA will reduce testing and redesign costs thereby shortening the product development time.
- Identify issues in designs before mold is made.
- Redesign components and systems before dependencies to other parts/components prohibit changes.
- Optimize design before prototyping.

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EXPERIENCE BASED CONSULTING IN PRODUCT DEVELOPMENT AND MATERIAL TESTING

- ➤ Material Evaluation
- ➤ Product Design and Analysis
- Material Characterization and Durability Testing
- > Failure Analysis
- Patent Development
- Feedback and Optimization of Rubber Compounds

PUBLICATIONS & REFERENCES

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